





### Flight **Demonstrations** for **Development** of a **Solar-Powered** Unmanned Aircraft











### **Dev Milestones**







# **Safety Piloted Vehicle**



- SPV allows reducing scope and design assurance
- Incremental approach to HW mods and SW features
- Flight functions are validated as available reducing risk
  - Program cost and timeline saving







#### • Flight Test as part of the engineering organization

- Involved in early System Design and Testing
- Direct engineering support of flight test
- Training
  - Reused legacy SI2 Pilot training
  - Table-top, simulator, low and high speed taxiing

#### Decision making

- Coordination with weather for window selection, airspace use and extending the flight window
- Coordination with engineering for flight test card selection
- IP-based comms network to the airplane

#### Autonomous system interaction

- AP drives control chain and yoke with bumpless engage
- AP increasingly used for airspace mgmt





#### Safety Boundaries:

- New functions are outside the safety boundary
- Reused SI2 FCC as OMC
- Surfaces reuse legacy SI2 safety boundary
- Throttle used a custom developed APG

#### Pilot Response to Failures

- Pilot reaction time in case of hardover •
- Limiting case is stall speed • severe limitation in elevator authority

2.6

2.4

2.2

2

1.8

1.6

4

Time to stall [s]

#### **PtFs**

- Incremental PtFs
- PtFs per HW mod



Max Deflection [deg]







#### • Challenges

- Wind and turbulence limits
- Upper-level jets in the Albacete region and sparse data
- Models
  - Mesoscale models for thermals and gusts
  - LiDAR: monitor wind and vertical stability
  - Radiosonde for models above and correlation
  - Allowed capturing nocturnal jets and thermals missing in global models

#### Weather windows

• Summarized probability matrix: factors over time

• Successful denlovment• 12/16 Winds Aloft Contour





Winds Vertical Profile from LiDAR (Mag/Dir)



Upper-Level Winds Contour Plot



# **Navigation System**



#### • INS/GNSS integration

Provides a tightly-coupled solution

#### • GNSS Installation

Good visibility, RAIM always active on INS and GNSS

#### • INS installation

- Assessed location for auto flight performance
- Verified location introduces no ASE issues

#### • Successful in air-realignment of the INS

- Completed under low dynamic conditions
- Readmit method increases availability needed to achieve 90-day flights (simpler than solution transfer methods)











#### • Challenges

- Low speed: 15 kts at TO/LDG
- High accuracy: ~2 kts diff b/w min power and stall

#### New ADS

- Multi-hole probe
- Modular probe-ADC

#### • Installation analyses

- Location chose to smooth correction changes
- Traded with mechanical performance and safety
- Installation errors corrected based on angles

#### Incremental updates

- Extended calibration data as necessary
- System ready for unmanned flight







-3

-2 AoA [deg]



# **Flight Control**



#### • Challenges

- High airspeed accuracy with efficient control action
- Size to speed ratio requires corrections  $(u_{\infty} \gg r \cdot b)$
- Longitudinal modes: fast due to low wing loading
- Lateral modes: adverse yaw, control coupling and little authority

#### • Fast Flight Controls progression

- May: loop servo closure
- July: attitude, airspeed and climb rate, initial track holding
- Sep: initial VNAV/LNAV waypoint nav, high wind protections
- Nov: complete waypoint nav, elevated approach, takeoff/landing shadowing

#### Incremental improvements

- Servo loop closure to alleviate stiction
- High wind protection to improve leg capture



Roll and AOS step responses







#### **First Waypoint Navigation engagement**



# **Progression of Flights**

organization



• Take-off and landing logic shadowing

NATO

OTAN

• +3 hours of AP engagement, 5 hours







# Thank you

